

RLSO 2

Robotic Lunar Surface Operations 2

2019

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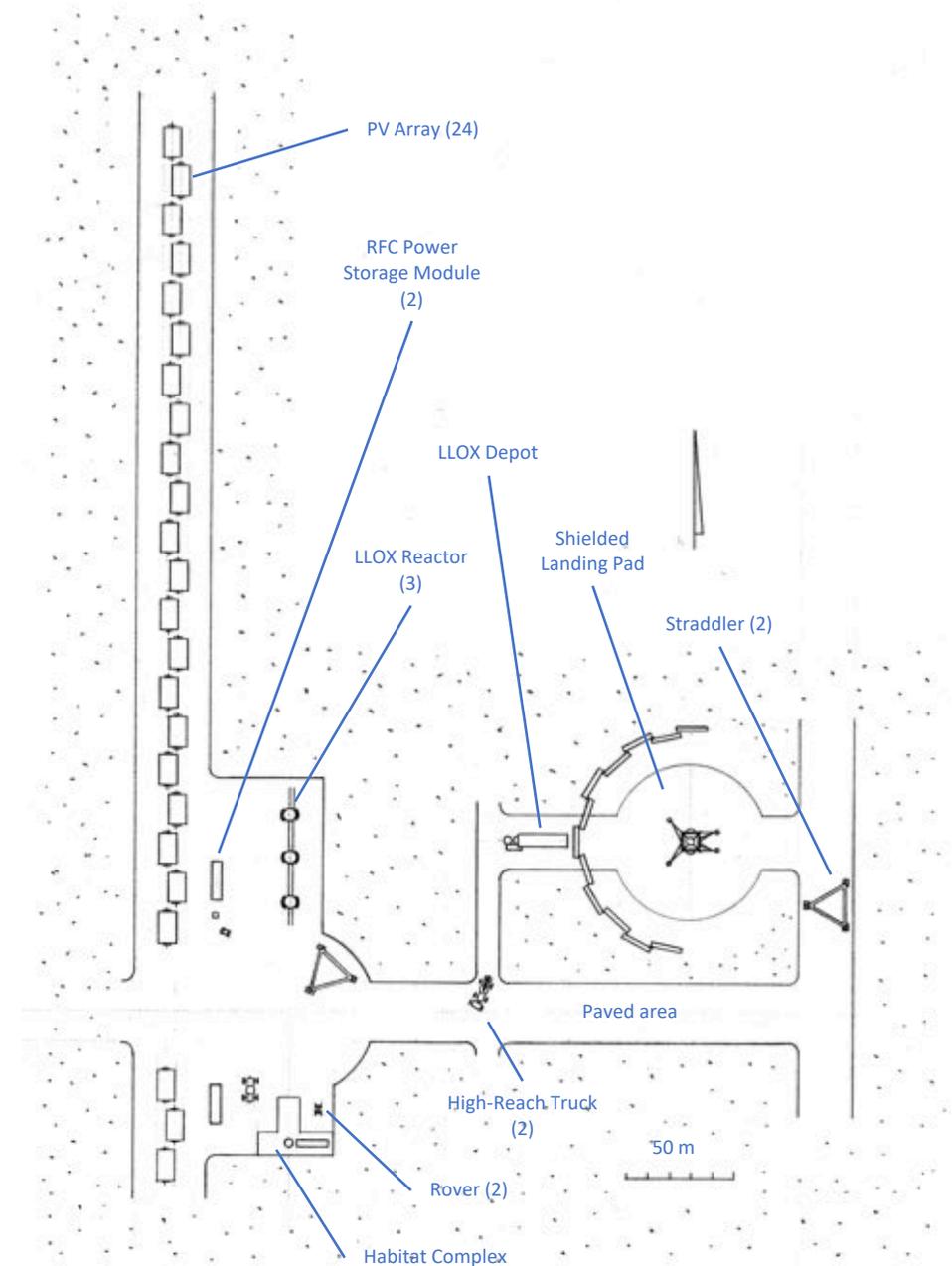
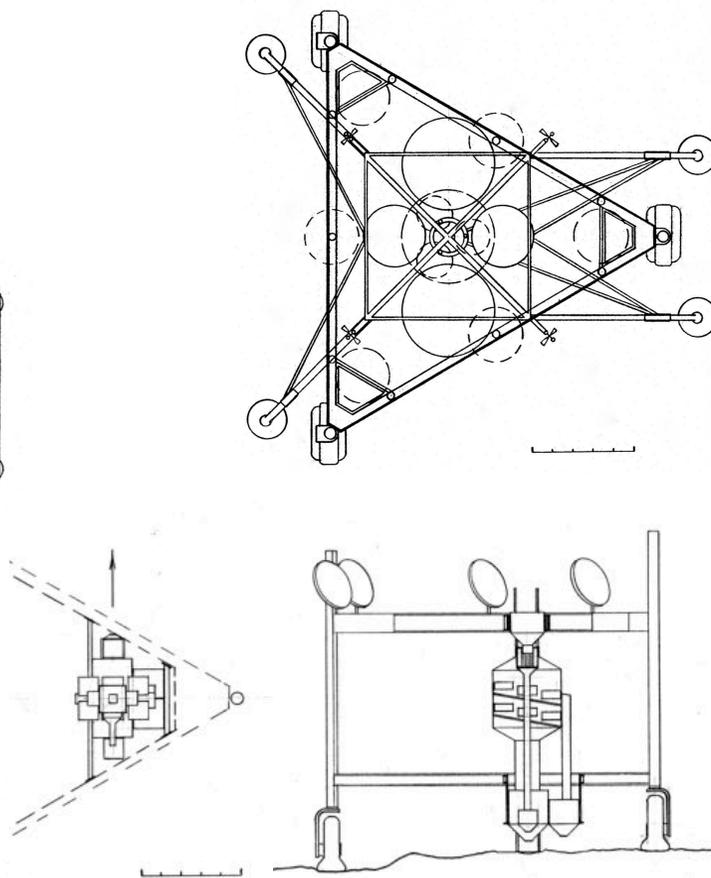
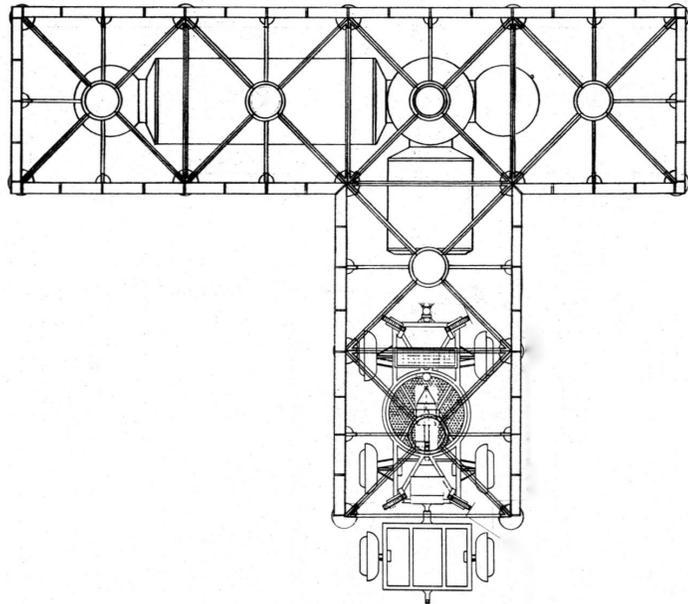
Jet Propulsion Laboratory
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RLSO – 1989 Boeing study for NASA ARC

“Develop a concept for a LLOX-producing lunar base that would be built by robots before human crews arrive”



Integrated element designs. Quantified operations analysis.



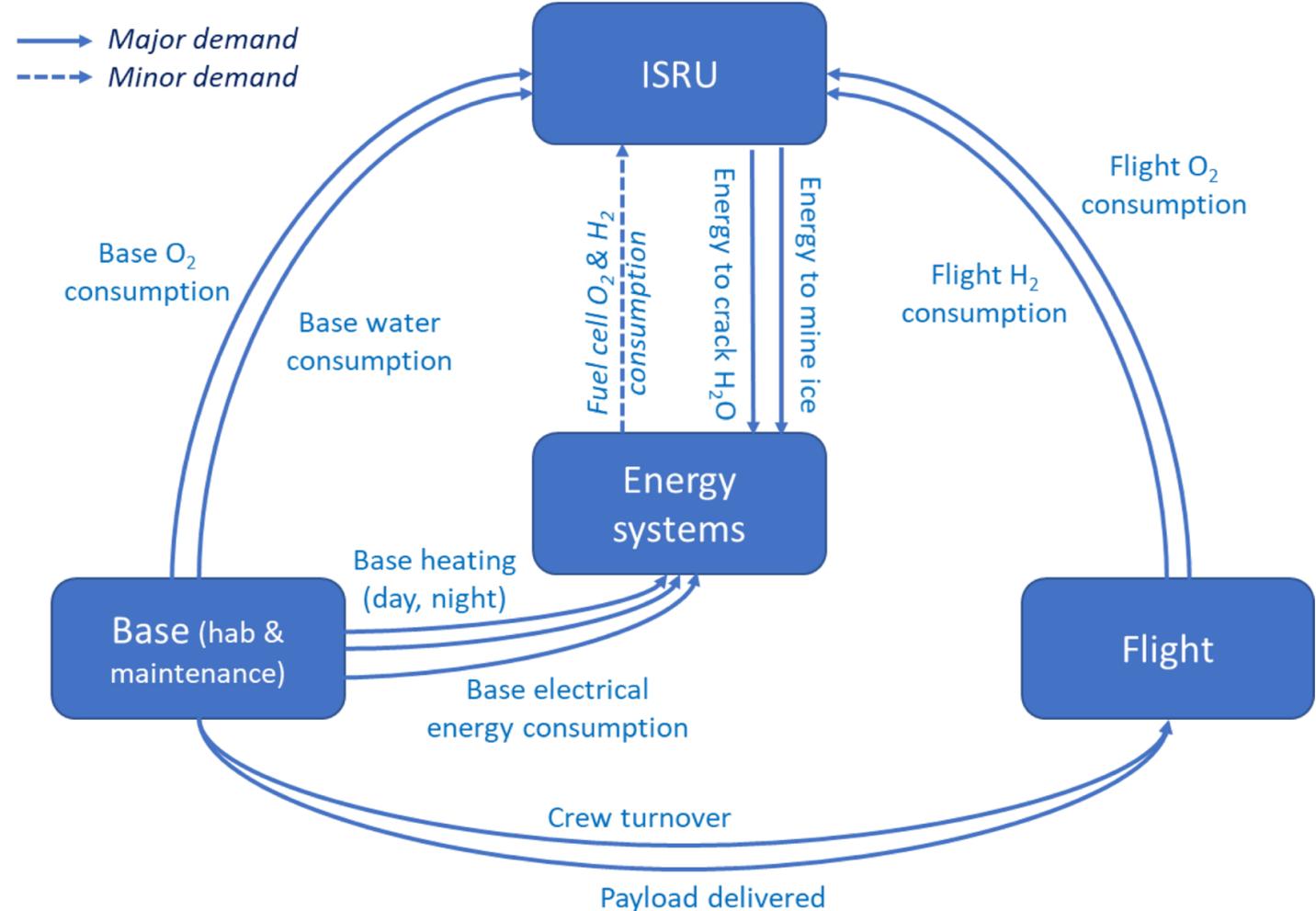
What would RLSO look like in 2019?

- Lunar polar volatiles, including ice
- ISS experience, international collaboration
- SPD-1, Moon Village, commercial actors, private capital
- SLS, Orion, Gateway, CLPS, Blue Moon
- Modern tools: spreadsheets, CAD, performance models

The importance of quantitative operations modeling

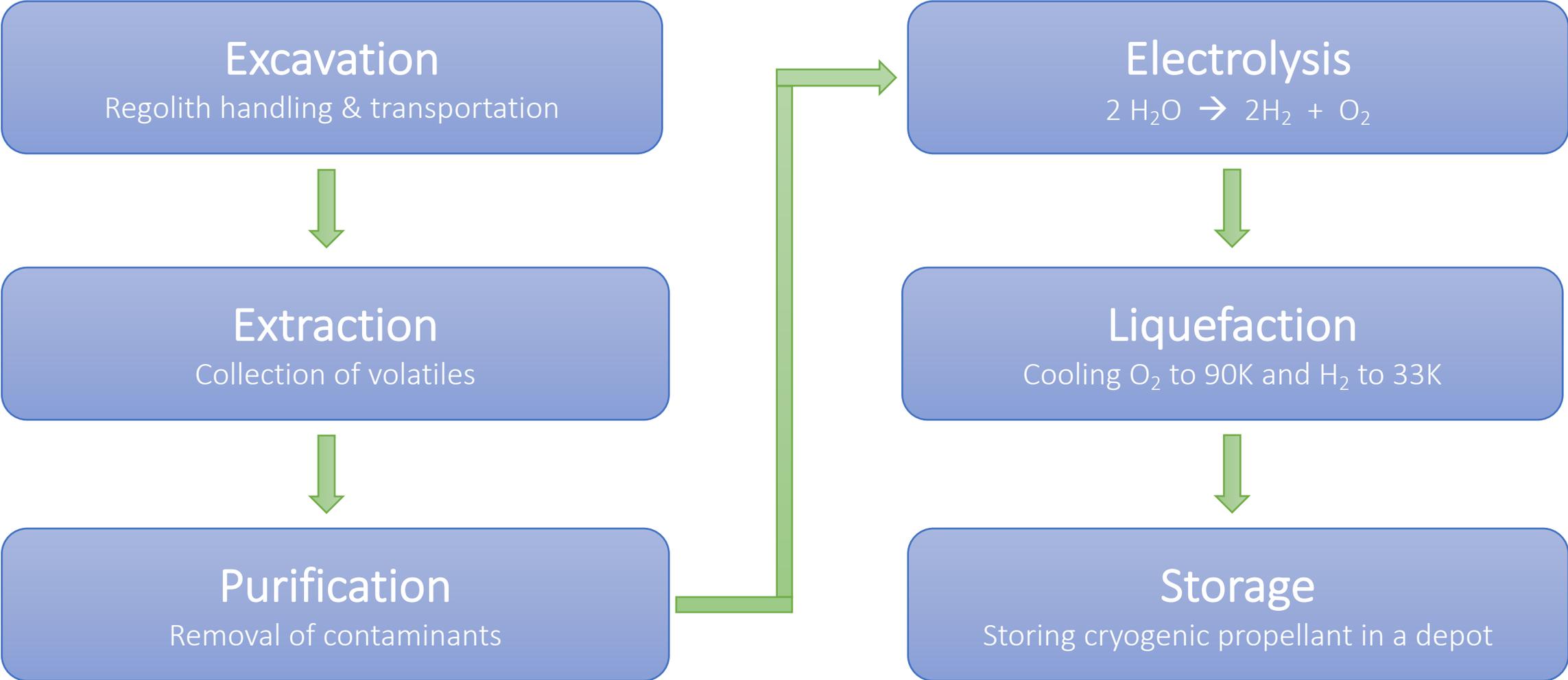
Each major element of the ops model is **flexible** and **expandable**, allowing for the integration of a wide variety of base element designs

- ISRU techniques and elements
- Energy system architectures
 - Lunar lander designs
 - etc.



The integrated model seeks to gracefully handle the **interconnected aspects** of the lunar base in order **to size the entire base system**

Functional decomposition of ice-based propellant ISRU



Major elements of an ISRU base

Energy System – >500 kW capacity, near-100% duty cycle, modular units landed intact, then connected via cables or laser

Habitat System – 30-day visits: hab, logistics, workshop, EVA, regolith-shield superstructure

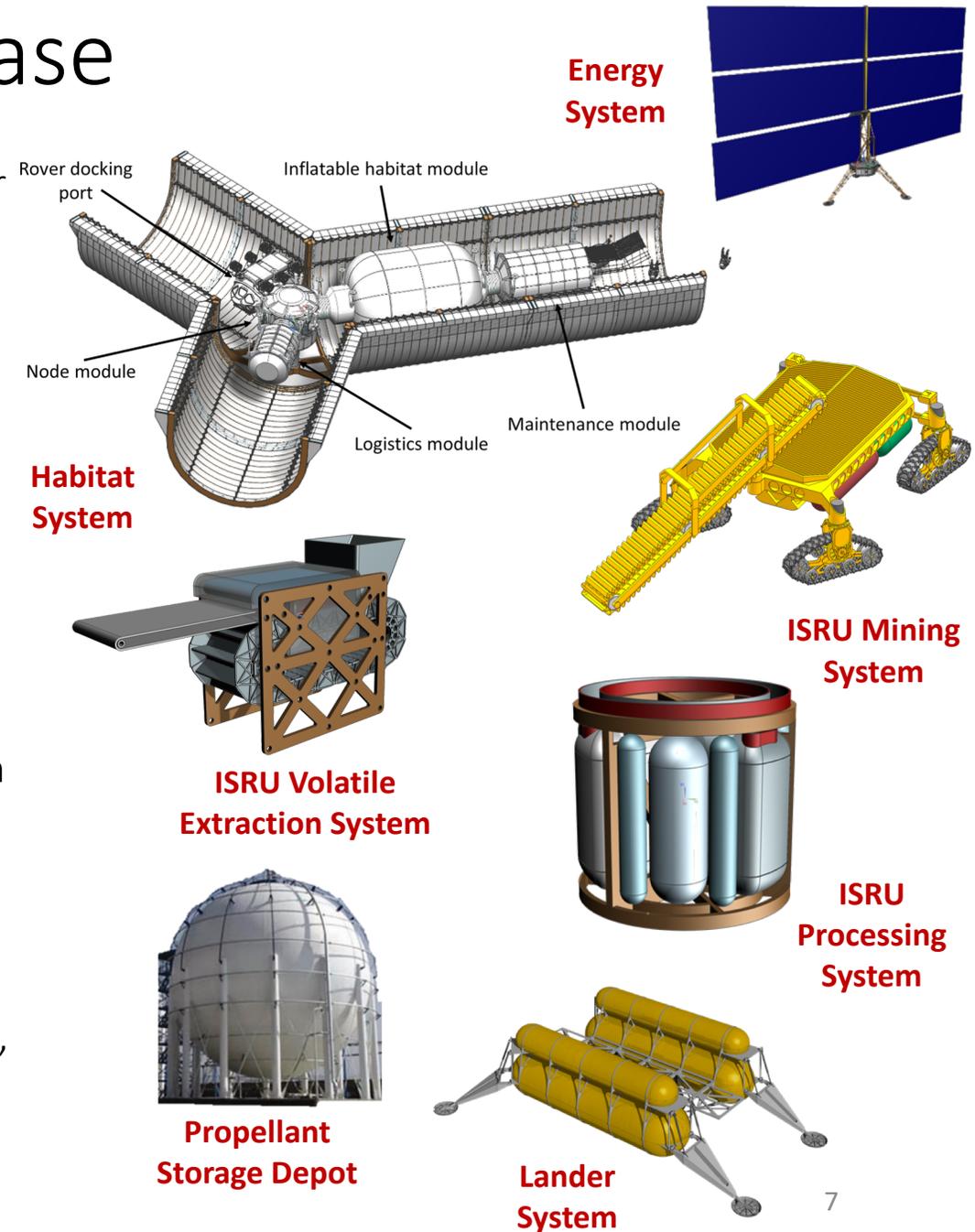
ISRU Mining System – Mobile robots that reach, excavate, beneficiate, and transport lunar regolith (or extract resource onboard and transport it)

ISRU Extraction System – Processor that separates frozen volatiles from lunar regolith

ISRU Volatiles Processing System – Plant that separates water from other volatiles, and cracks it into H₂ and O₂

ISRU Depot System – Plant that liquefies, cryogenically stores, and distributes cryogenic propellant to reusable landers

Lander System – Reusable, refuelable lander, reusable landing pad, and ground support systems

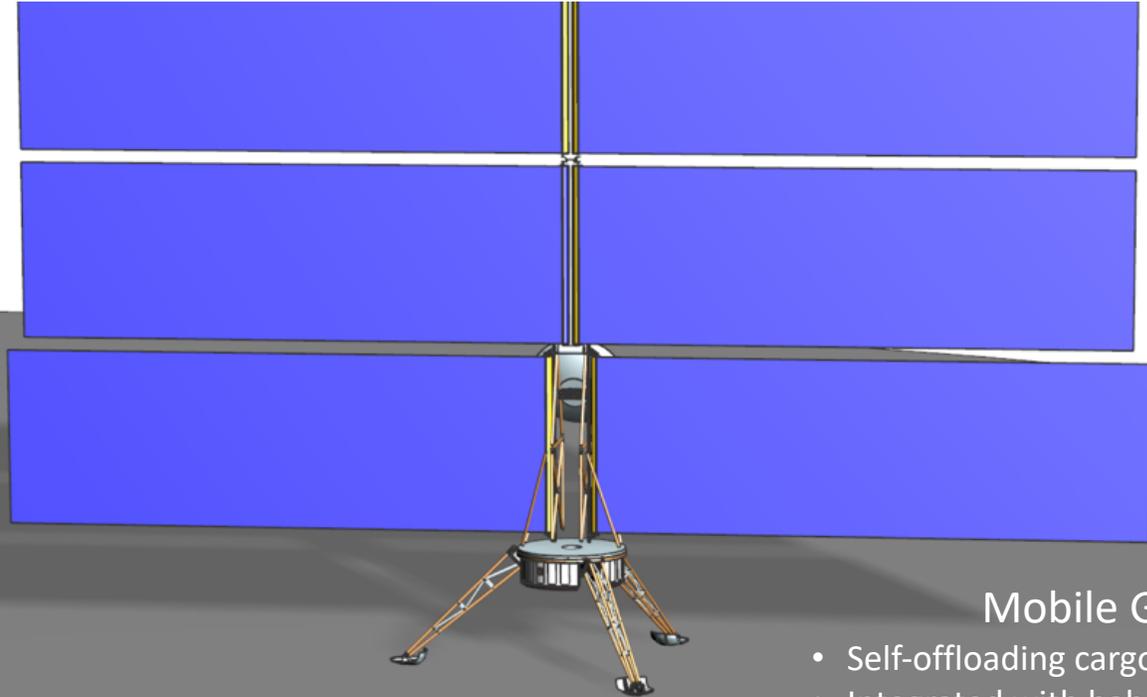


Base element designs must be integrated

PV Power Plant

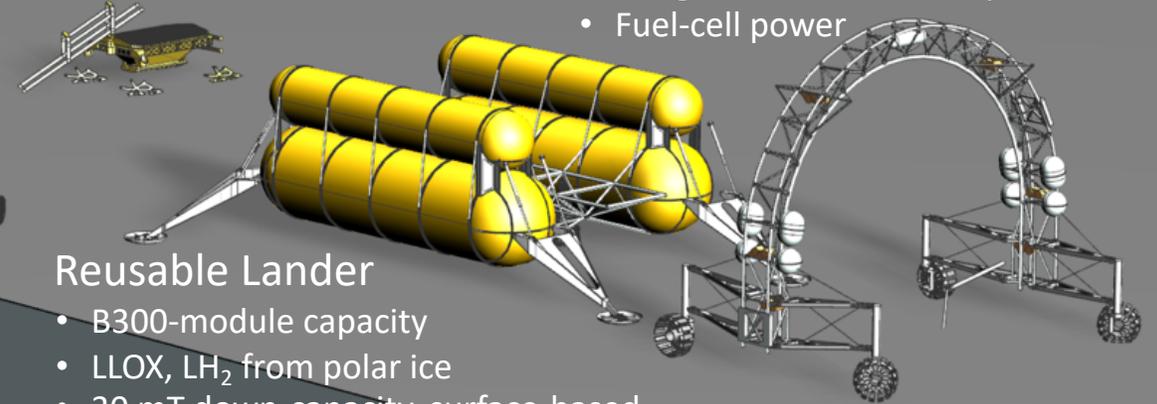
- 188-kWe BOL, modular unit
- 4 T, self-deployed
- Active area $\geq 4\text{m}$ above ground
- Compatible with Blue Moon delivery

10m



Mobile Gantry

- Self-offloading cargo handler
- Integrated with hab complex assembly
- Fuel-cell power

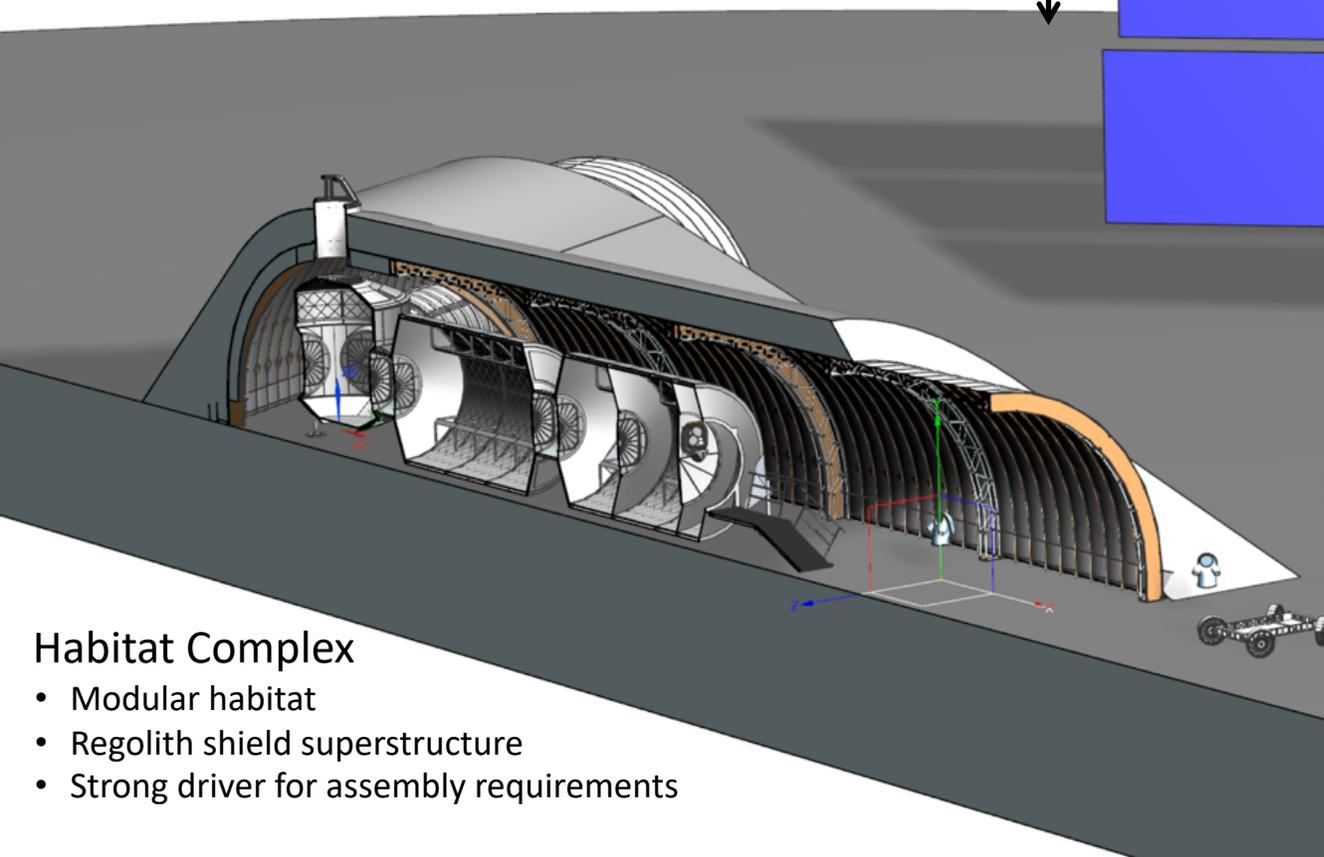


Reusable Lander

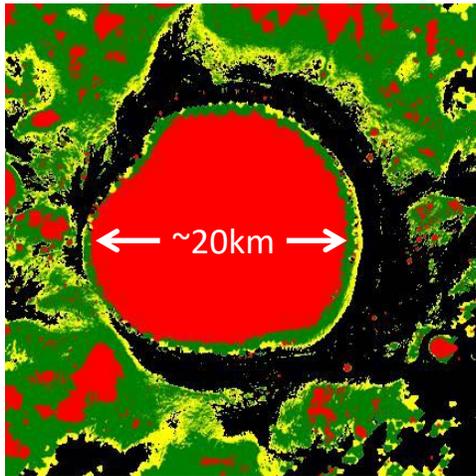
- B300-module capacity
- LLOX, LH₂ from polar ice
- 30 mT down-capacity, surface-based

Habitat Complex

- Modular habitat
- Regolith shield superstructure
- Strong driver for assembly requirements



Polar ice resources



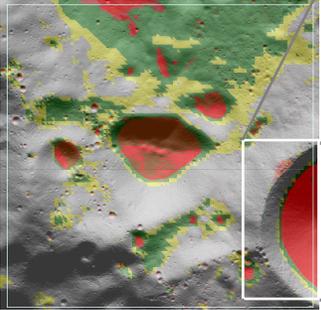
- Type 1
- Type 2
- Type 3
- Type 4

- Bin by water-stability depth into four terrain types
- Map areas that have 20-m DEM and high-res thermal models
- Illustrated: Hermite-A crater, lunar north pole

	Total Fractional Area (%)	Water concentration (wt%)	Depth beneath the surface (cm)	Water-containing column (cm)	Total water excavated (kg/m ³)	Extraction area for 10 t of water (m ²), @30% patchy
→ Type 1a PSR regolith	9	2	20-100	80	7.2	1,400
Type 1b PSR surface frost	9	100	0 - 0.002	0.002	0.006	> 1.5M
→ Type 2 PLR buried regolith	28	1	40-100	60	2.7	3,700
Type 3 PLR deeper regolith	7	0.5	60-100	40	0.9	12,000
Type 4 Lunation-lit regolith	56	0	--	0	0	n/a

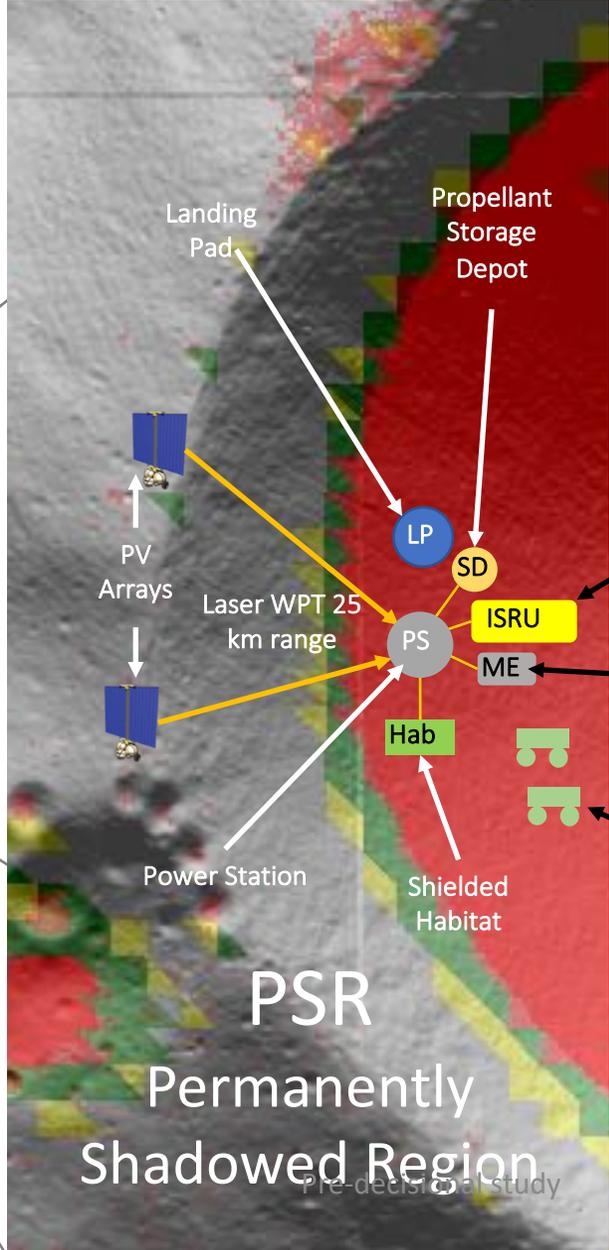
Option 1 – Deep Shackleton, PSR

Type 1a Resource
 2 wt% water ice, found
 20 – 100 cm down

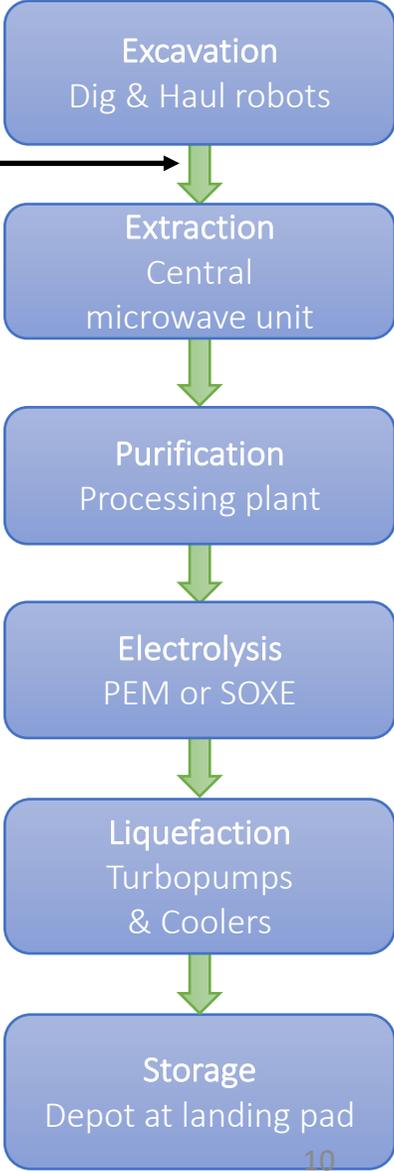
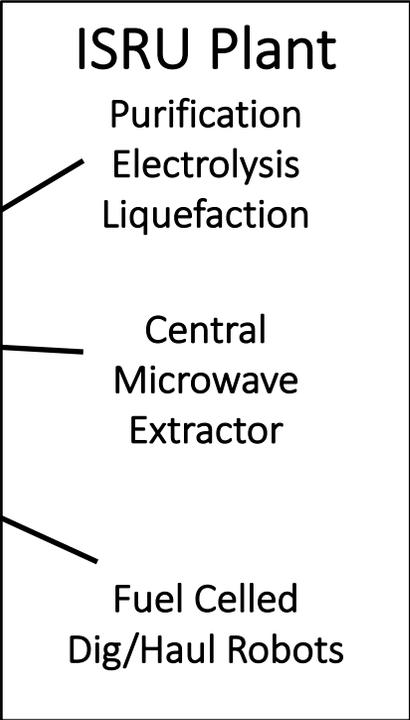


Power Infrastructure

- Multiple PV rim stations yield high lunation duty cycle
- Laser WPT to central power station
- Cable distribution to base elements
- Mobile elements use fuel cells, recharge at central station

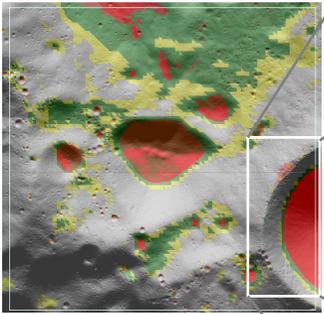


Reach, remove,
 and haul
 regolith resource
 <1 km to ISRU base

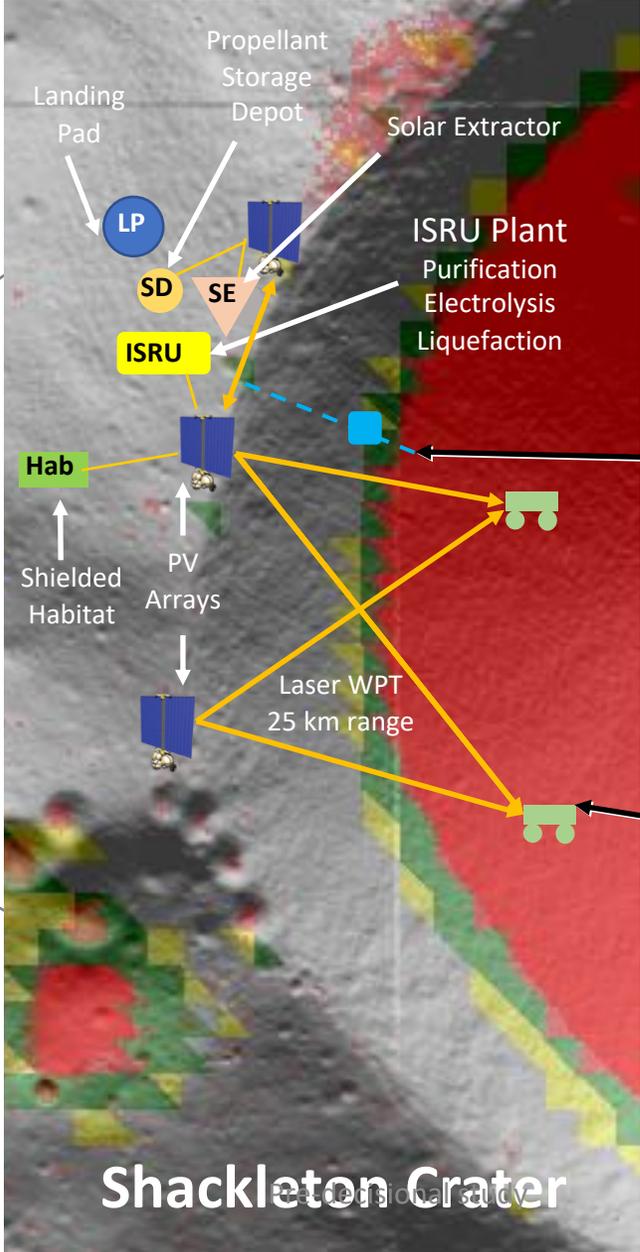


Option 2 – Shackleton Slope, into the PSR

Type 1a resource
 2 wt% water ice, found
 20 – 100 cm down



- Power Infrastructure**
- Multiple PV rim stations yield high lunation duty cycle
 - Power cables to base elements
 - Laser WPT to excavators inside PSR
 - Fuel-celled base robots



Haul benefited resource <10 km up and out of the crater

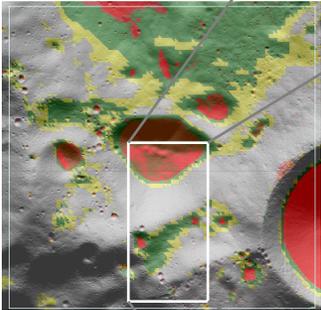
Resource "escalator"

Beam-powered Roving Beneficiators

Option 3 – Shackleton West Ridge, PLR Ice Fields

Type 2 resource

1 wt% water ice, found
40 – 100 cm down

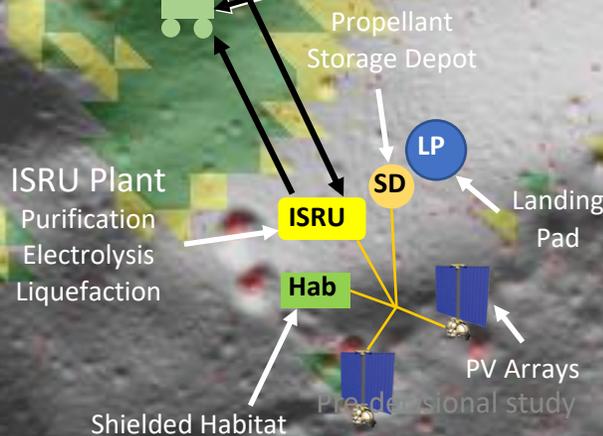


Power Infrastructure

- Multiple PV stations yield high lunation duty cycle
- Solar/fuel cell mobility
- Excavator Extractor Retriever and base robots

PLR
Persistently Lit Region

Rovers retrieve
volatiles to base
(3-8 km)



Transport frozen volatiles
to base (3-8 km)

Fleet of resource rovers

- Core into the buried resource
- Heat the cores in situ
- Freeze the volatiles
- Return to base

Excavation + Extraction
In situ extraction
by coring rovers

Purification
Processing plant

Electrolysis
PEM or SOXE

Liquefaction
Turbopumps
& Coolers

Storage
Depot at landing pad

Initial ISRU Requirements

Water need

- Lander flights per year: **4**
- Propellant required per flight: **40,000 kg**
- Water required per flight: **51,500 kg** (6:1 engine ratio vs. 8:1 water mass ratio)
- Water need: 206,000 kg/yr = **1,130 kg/d** @ half-time operations

Resource assumptions

- Type 1: **0.15 m³** (~210 kg) regolith per kg of H₂O yield
- Type 2: **0.40 m³** (~600 kg) regolith per kg of H₂O yield

Regolith need

- Type 1: **240,000 kg/d** @ half-time
- Type 2: **680,000 kg/d** @ half-time

Base Energy Requirements

Volatiles processing minimum: **10.5 kWh per 1 kg of water**

- 2 kWh/kg for extraction from regolith
- 6.5 kWh/kg for electrolysis: H₂O into H₂ and O₂
- 2 kWh/kg for liquefaction: H₂ and O₂ into LH₂ and LOX

ISRU energy: **2,200,000 kWh/yr**

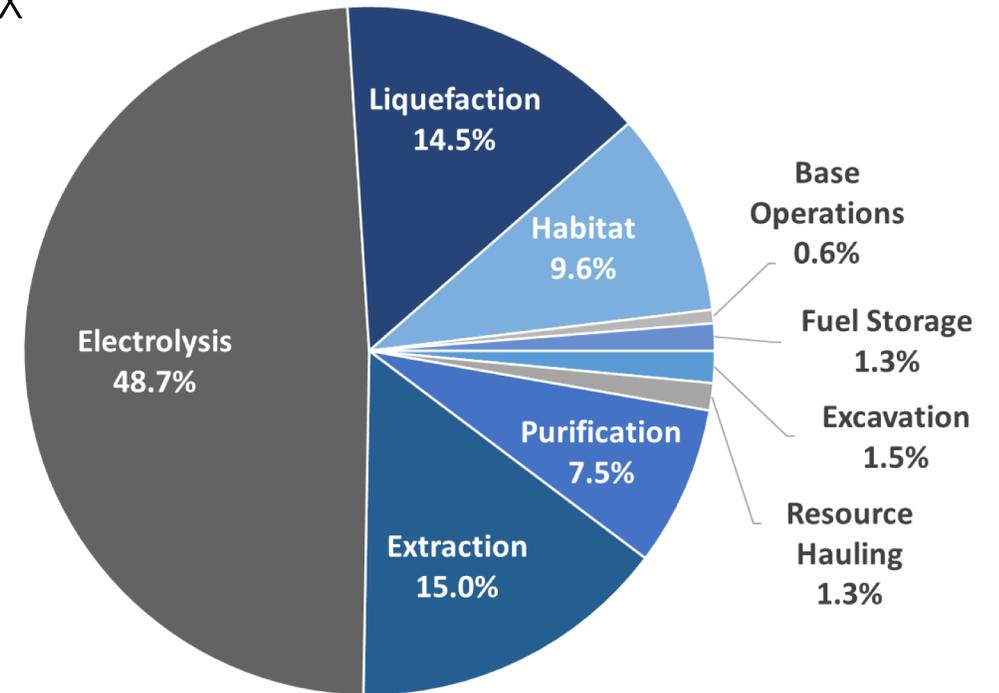
- Quantity of water required: 206,000 kg/yr

ISRU power: **500 kW @ half-time ops (4,380 hr)**

Other energy requirements include:

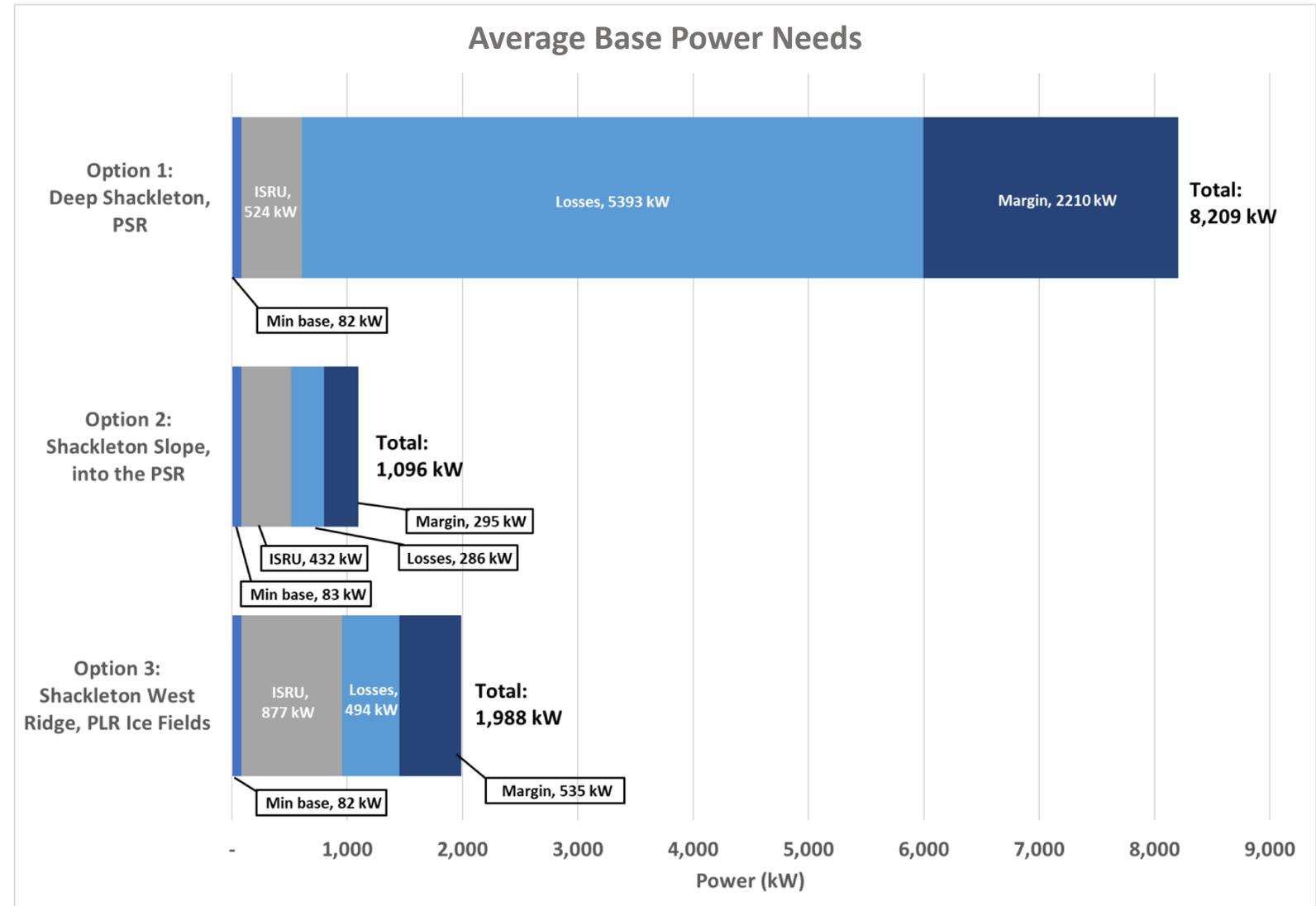
- Excavation, hauling, cryogenic storage
- Hab-complex sustained operation
- General mobility and base operations
- Power losses (cables and beaming)

Total base power need: **≥600 kW**



Comparing energy needs across base options

- Minimum base (habitat + flight) power need is approximately the same for each architecture
- ISRU dominates power need over minimum base
- Cable and beaming losses are a substantial fraction of the power budget in all cases
- Scheme 1 suffers significantly higher losses because the energy-expensive processing is a long way from the power source



Option comparison

Option	Best attributes	Worst attributes
<p>1. Deep Shackleton</p>	<ul style="list-style-type: none"> • Best quality resource, with minimal overburden removal • Stable operating environment: dark, 70K 	<ul style="list-style-type: none"> • Significant power distribution losses • Base cannot easily access regional exploration sites • Base is permanently dark
<p>2. Shackleton Slope</p>	<ul style="list-style-type: none"> • Best quality resource, with minimal overburden removal • Base can support exploration excursions 	<p>Resource must be brought several km up and out of crater</p>
<p>3. Shackleton West Ridge</p>	<ul style="list-style-type: none"> • Avoids crater slopes • Proximate sunlight and shadow • 0.5m/px LROC imagery • Base can support exploration excursions 	<p>“Half-quality” resource, buried deeper</p>

Emergent findings

“Best” ice resource and location may not be in a PSR

Nuclear power useful for production-scale ISRU would have to be MWe class

Potential competitive roles for commercial actors

- Power providers, extraction rovers

Empirical knowledge gaps with high leverage

- Vertical distribution at m scale – wt% of ice as a function of depth
- Horizontal distribution at km scale – patchiness of resource “field”
- Geotechnical properties – “coffee grounds and sugar” or cryo-permafrost
- Diffusion rate – trapping vs losing the resource from heating in situ
- Agitation loss coefficient – losing the resource from handling it

